

AVIATION

The Oldest American Aeronautical Magazine

FEBRUARY 16, 1925

Issued Weekly

PRICE 10 CENTS



On the way to Hawaii—PN7 (Wright Tornado) scout plane in the hold of U.S.S. Wright

VOLUME
XVIII

SPECIAL FEATURES

NUMBER
7

THE NEW WRIGHT CYCLONE ENGINE
\$22,000,000 MORE FOR AIRCRAFT CARRIERS
CALENDAR OF THIS YEAR'S NATIONAL AIR RACES
DESCRIPTION OF SIKORSKY TWIN-ENGINED TRANSPORT

GARDNER PUBLISHING CO., Inc.
HIGHLAND, N. Y.
225 FOURTH AVENUE, NEW YORK

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under Act of March 3, 1879.

Ten Years of Packard Pioneering 1915-1925 in Aircraft Motor Development

See *For a review of Aviation History*



Packard Model 1



Packard Model 4



PACKARD'S pioneering in aircraft motor development dates back to the spring of 1915, when work on an airplane motor for war purposes was given precedence over all else in the great Packard factory.

To forget, as time passes, the part a mere corporation played in such a development, is a human weakness. It is more natural, perhaps, to be oblivious of the post-war progress made by the same corporation in its service to aviation.

Packard exists today, as it did ten years ago, a leader in the design and manufacture of aviation engines, known the world over.

PACKARD MOTOR CAR COMPANY
DETROIT, MICHIGAN

and the new 1925 model

FEBRUARY 16, 1925

AVIATION

VOL. XVIII NO. 7

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DAYTON, OHIO

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The Sikorsky S29A Twin-Engine Transport Plane

By ALEXANDER KLEMIN

Assistant Professor of Aeronautics, New York University

At the request of the Sikorsky Aero Engineering Corp. of New York City, a series of tests were recently carried out on the Sikorsky transport airplane by the women aeronautical students of New York University, under the writer's direction. Tests were made at Roosevelt Field, L. I.

Thanks are due to the Pioneer Instrument Co., who pro-

vided on each side of the cabin. As shown in Fig. 2 the doors are in two parts. The upper part is a sliding door and can be opened in the air. The lower part when opened provides a stairway leading down to the cabin floor. Passengers can communicate with the pilot through a door at the rear of the cabin. The cabin is free of all leads, girders, beams, etc.



Front and side view of the Sikorsky S29A transport plane, fitted with two 400 hp. Liberty engines

vide the use of their excellent instruments, thus greatly facilitating the tests.

Inverse weather conditions prevented the carrying out of the full program of tests scheduled, but sufficient data was secured to give an idea of the performance of the plane.

General Characteristics

Shown is shown a front view of the Sikorsky Passenger plane and also a side view. The general appearance, while not departing greatly from conventional lines, is clean-cut and good.

The pilot is placed in a snug cockpit and has excellent views in all directions. Instruments in the cockpit and the engine controls are well arranged. There is plenty of room for the navigator or observer who sits next to the pilot, and communication between them is easy. The number of the pilot and observer is well taken care of, an important feature in a large multi-engine plane.

The passenger cabin is roomy and comfortable, with splendid views through the windows, plenty of head room and room to move about. The ventilation is excellent. Doors are pro-

vided on each side of the cabin. As shown in Fig. 2 the doors are in two parts. The upper part is a sliding door and can be opened in the air. The lower part when opened provides a stairway leading down to the cabin floor. Passengers can communicate with the pilot through a door at the rear of the cabin. The cabin is free of all leads, girders, beams, etc.

Nothing could be more deplorable than a passenger's point of view.

Power Plant

The power plant is well installed. The motor installation forms with the oil installation a separate unit which can be removed and replaced in a very short time. The fuel system, consisting of four tanks and gravity tanks, is simple and reliable. The five tanks are reduced to a minimum. The fuel tanks are placed at the back of the engine nacelles and can be moved up and down to keep the balancing action, with a decrease in resistance when down and out exposed to the air.

In the central part of the cabin there are provided two springs providing the mechanism to step out on the wing and at one step to reach the motor in flight.

Structural Features

The structure of the plane is rugged and simple. It is particularly pleasing from a production point of view. The main spars of the wing are made up of subaluminum

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and channels, forming I-beams. A number of stringers are spaced about 2 ft. and made of duralumin channels and are 6.025 to 6.625 in. thick, except the two main spars with an extra space running parallel to the main spars. Light sub-aluminum ribs of 0.020 in. gauge channel section are placed in the main and false spars. The general character of the wing structure is illustrated in Figs. 1 and 4.

Figs. 2 and 3 illustrate the construction of the fuselage. The framework of the fuselage are made of steel angles. At the forward section of the fuselage the struts are made of

tubing of adjustable stabilizer in thrust line, —2 to +4 deg. Sudders (13 balanced), total area 38 sq. ft. Weight empty (with water), 7,775 lb. Specified useful load, 4,225 lb. Specified gross weight, 12,000 lb. Wing loading, 32.1 lb./sq. ft. Power loading, 13.2 lb./hp. Power plant, 2 Liberty motors rated 400 hp. at 2100 r.p.m. Cabin space, 26 ft. by 4 ft. by 4 ft.; total volume, 416 cu. ft.



Construction details of the Sikorsky transport plane—1, wing structure; 2, Mr. Sikorsky in the passenger cabin in process of construction; 3, engine mounting; 4, portion of wing spar; 5, fuselage structure

and channels, and toward the tail of duralumin channels. The intensive bearing of the cabin is very carefully taken care of by great pieces as shown in Fig. 6.

The engine mounting as shown in Fig. 3 is extremely strong and well designed.

The landing gear, with shock absorbers hidden in the wing, has an exceptionally large shock absorber travel (12 in.) and gives very little bump on landing. It is well braced in all directions.

Control surfaces are also built up of steel and duralumin in practical and reliable fashion.

Control Surfaces

The control surfaces are simple and easily operated. An adjustable stabilizer is provided, which is readily operated by the movement of a lever in the pilot's seat.

The two outer rudders are single connected with the rudder as the inner side. This is particularly useful in the case of fold with one motor. When both motors are on, the rudder is moved, position stabilizer one rudder. When one engine is throttled down or goes out of commission, the rudder in the direction of the motor still running will ensure a turning tendency opposing the turning tendency of the motor. This simple arrangement should prove very useful for flight with one motor.

General Data

Overall length, 49 ft. 10 in.
Overall height, 13 ft. 6 in.
Span, upper wing, 40 ft.
Span, lower wing, 32 ft.
 chord, upper wing, 10 ft. 3 in.
 chord, lower wing, 8 ft.
Wing surface equipped, Sikorsky 18 (modification of Götting 18)
Incidence in thrust line 4 deg.
Total area of wings (including ailerons) 992 sq. ft.
Area (12 on upper wing) 72 sq. ft.
Forward stabilizer area, 38 sq. ft.
Lift area 50 sq. ft.

Designed safety factor, high incidence, 5; low incidence, 4
Propeller: Hamilton
Diameter, 30 ft. 4 in.
Pitch at 1/2 radius, 6 ft. 10 in.

Performance Tests

Flight test was Motor Liberty—flight and lift load tests were made at a benching altitude of about 1200 ft. with right motor at 2100 and left motor throttled down to 785 r.p.m. Both right and left hand tests were made with one, but no steep banks were attempted.

| | |
|---|---------------|
| Climb with one Motor Liberty | Dec. 16, 1924 |
| Weight empty (with water) | 7,775 lb. |
| 9.0 sec. climb | 10,500 lb. |
| 100 psi. gas at 6 ft. | 6,615 lb. |
| 9.0 sec. climb at 7 ft. 10 in. | 47 lb. |
| Equipment included 2 starters, battery, 10 chains, fire extinguishers and misc. | |
| Incidence | 241 lb. |
| Total fuselage load | 2,485 lb. |
| Gross weight | 10,261 lb. |
| Initial recording barograph reading, 900 ft. | |
| Final recording barograph reading, 1200 ft. | |
| Direction of climb on one motor, 15 sec. | |
| Temperature, 28 deg. F. | |
| R.p.m., right motor, 2540. | |
| R.p.m., left motor, 750. | |
| Average air speed during climb, 20 mph. | |
| Corrected climb at standard air at 1225 ft., 26.4 ft./sec. | |
| Maximum Speed, Climb and Ceiling | Jan. 5, 1925 |
| Weight empty (with water) | 7,775 lb. |
| 12 sec. climb | 2,485 lb. |
| 22 psi. gas | 700 " |
| 22 psi. gas | 60 " |
| Equipment | 165 " |
| Disposable load | 2,892 lb. |
| Gross weight | 10,657 lb. |

Maximum speed converted to standard air at 50 ft. altitude, 112.2 m.p.h.
Minimum speed, 58.8 m.p.h.
Initial rate of climb, standard atmosphere, 692 ft./min.
Service ceiling, 12,500 ft.
Time of climb to 5,000 ft., 8.5 min.
Time of climb to 10,000 ft., 22 min.
Maximum air draft up to 10,000 ft., 18,000, L. 10,000.
Minimum air draft above 10,000 ft., L. 2,500; L. 1,400.

Pilot and Observer's Reports

The pilot during the performance tests was Roger L. Sholkey, designer of the plane. Student observers were John Bonbrake and O. Leland.

Both student observers and the writer are able to report that the engine of the airplane is completely new. Several test conditions in flight were observed under severe weather conditions. Outstanding flight and power were with one motor throttled off and no difficulty in the pilot. Landings were always made with ease and comfort. Get away was rapid.

Book Reviews

Dr. GLENN T. ANDERSON, DASH RAYSON, By André Lefebvre, Pilot-Officer of the Reserve, Professor at l'École Supérieure-Claix, Paris (120 pp.), profusely illustrated. Gauthier-Villars et Co., Paris.

This is a second, revised edition of an excellent text book for airplane pilots which has become a classic in France.

Dr. YVES A. VALLÉE, BREVETÉES EN CHIMIE, By Louis Breguet, 400 pp., 1924. Gauthier-Villars et Co., 55 Quai des Grands-Augustins, Paris, France.

In this book M. Breguet gives a mathematical analysis of that little known phenomenon which the late Professor Langley called the "internal work of the wind" and which numerous experimenters are attempting to define in the operation of winged machines. The author states that about all we know of this phenomenon is what Langley discovered, and that later experimenters generally confirmed his findings without, however, materially adding to the knowledge available.

DR. ROBERT L. COMPTON, By Pierre Kellerman (70 pp., 60 ct.). Verlag von M. Kreyer, Berlin W. Germany.

The article and the development of the rapid type of airplane, according to Count Ferdinand von Zeppelin from 1900 on down to the present days is readily described in this work which is of the well-known, popular type.

Coming at the same time as the highly technical work on Zeppelin aeroplanes by Pierre Besson, Pierre Kellerman's book fills a long felt need for an up-to-date compendium on this subject which the average person interested in lighter-than-air craft can understand, without being an engineer. The author describes the various chief types and the progress of the design of each step and follows this up with brief notes giving the history of each step. A large number of photographs, illustrations and tables are given, a useful index.

REPORTS A3N, A3B AND A3T OF THE NATIONAL ADVANCED RESEARCH BOARD (N.A.R.B.).

Report A3N concerns pressure distribution experiments on the fore-edge of an airplane model.

Report A3B contains data for the correction of air-cooled drag of model airfoils tested in an anemometer of this department.

Report A3T deals with experiments on the pressure drag on an air current made by model wings.

THE MECHANISM OF FLUID VELOCITY AND PRESSURE, By J. R. Froude, (136 pp., 45c). Edited by Arnold & Co., London.

The subject of this book was a member of the British Admiralty Committee for investigation who met his death in the disaster which befell the rapid aircraft H22 in August, 1931. He was a specialist on all problems connected with aeroplanes, and he is noted in England had such research on aeroplanes.

Level C. E. Archer, A. E. McCook Field, reports on the following as follows:

"It is very easy on the contrary, a very slight movement of the wheel being enough for straight flight or to make the day of this flight, Nov. 15, 1933, the air was quite rough. I was particularly interested in the rigidity of the structure in flight. There was not the slightest vibration. It was just that I could detect and the motion was quite regular. The airplane flew and actually climbed about 150 ft. in a very few minutes on one motor. The take-off was made in 9½ sec."

A Good Commercial Design

The Sikorsky plane shows that it was built under most ideal conditions, with very little shop equipment involved. Although adequate strength was maintained throughout the design in the amount that the weight could be such as to be by 700 lb. the actual weight was only 1,000 lb. when it was about 7,000 lb., and the disposable load 5,000 lb. The design, while capable of refinement and lightening up, is an excellent commercial job suitable for passenger or freight transportation.

At the time of his death he was completing a treatise on the measurement of fluid velocities and pressures, a subject to which he had devoted the last years of his life. As a tribute to his memory this manuscript was completed and edited by E. A. Froude, a colleague most closely associated with him in his research work.

The book is divided into the following chapters: I. Principles. II. Instruments. III. Flowing Part Instruments. IV. Flowing Wire Anemometer. V. Duct Flow and Velocity Measurement. VI. Flow of Fluids in Conduits. VII. Flow of Fluids in Open Channels.

ANALYTIC MECHANICS, By Dr. Joseph R. Ames, Ph.D., I.S.D., Chairman, Engineering Department, S.A.C. (Oxford, Ontario, Canada, Publication 2728, Washington, D. C.).

A brief outline of the subject of mechanics, mechanics and of the field in which it can most profitably work in America.

ANALYSIS OF FLUIDS AND LIQUIDS, FROM 1905-1930 (60 pp., 7 charts). (Gauthier-Villars & Co., Paris, France).

The "Bureau des Longitudes" section in France mark the same progress which the Bureau of Standards does in the country. The annual under review contains a considerable amount of statistical information on the world's population, meteorology, astronomy, weights and measures, the principal countries, etc.

Forces on Airships in Gusts

NACA Report No. 204

The trials of the Shenandoah have proved that, as we previously suspected, the aerodynamic loading moments in appreciably straight flight in gusty weather conditions are much more severe than those which would be expected. It has been assumed that the conditions encountered in gusts could be approximately represented by assuming the airship to be in an air current angle of α degrees (according to whether the gust is horizontal or vertical), the instantaneous angle being $\alpha + \frac{1}{2} \sqrt{V}$, where α is the component of the velocity of the gust at right angles to the longitudinal axis of the ship, and V is the speed of the ship.

In this report, prepared for publication by the National Advisory Committee for Aeronautics by C. F. Branson it is shown that in determining the instantaneous angle of pitch or yaw the acceleration of the gust is as important as the maximum velocity. An expression is derived for the instantaneous angle in terms of the speed and certain aerodynamic characteristics of the ship, and the expression for the longitudinal axis of the ship, and V is the speed of the ship and the acceleration of the gust, and the application of the expression to the determination of the forces on the ship is illustrated by numerical examples.

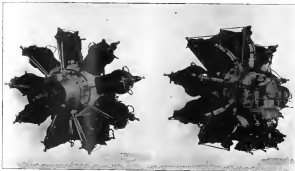
Report No. 204 may be obtained upon request from the National Advisory Committee for Aeronautics, Washington, D. C.

The Wright Cyclone Aviation Engine

New 400 hp. Air-cooled Radial Successfully Passes 50 hr. Navy Endurance Test

"Slightly more than a year ago, the Bureau of Aeronautics of the Navy Department felt the need of an air-cooled powerplant to replace the Liberty engine in certain types of ships, principally torpedoes and observation planes. This requirement

plunged the crankshaft with the exhaust valve forward and the inlet valve to the rear. The operation of the exhaust valve is controlled by means of a push rod and rocker arm, but the inlet valve is operated by a pull rod across the top of the



Front and rear views of the new 400 hp. Wright "Cyclone" radial air-cooled engine

was met by the development of a powerplant of lower displacement than had been considered in this country before. To add to the difficulty of designing such an engine, a limitation was placed on the diameter in order to keep the parasite resistance

as low as possible. A contract was given the Wright Aeronautical Corp. of Dayton, N. J., for the development and construction of three test engines, into which have gone the combined experience of the Lawrence as well as the Wright Companies. In this connection, it is interesting to note that the Wright Company, in 1920, won an Army design competition for a 300 hp. air-cooled radial engine, and subsequently built three engines, known as the Model B-1, which were the first large air-cooled engines to be built in the United States.

Dimensions of Engines

The Wright "Cyclone" or P-3, as the new engine was called in the original engineering plans, is a four-cylinder, with two cylinders having a 4 in. bore and 5½ in. stroke. The total displacement is 1,800 cu. in., which is the same as that of the Liberty engine.

The external appearance of the engine, as can be seen from the accompanying illustrations, is remarkably clean. There is no complex framework that will be necessarily removed, namely, the valve gear. The valves, instead of being located in the plane of right angles to the crankshaft, are located in the

valves. The valve gear is located in front of the engine, and all the accessories at the rear, including carburetors, magneto, pumps, etc.

The cylinders have steel sleeves, steel fins and aluminum heads. The development of this engine followed the usual practice of the Wright Company, namely, the design and development of a single cylinder, followed by the complete design of the engine and the construction of the first one. It was found that the original conception of the engine was detrimental to the proper operation of the engine, so that it was necessary to increase the length of the cylinder slightly, although even with this slight increase the diameter of the engine from the top of the cylinders is very much less than any engine of this power heretofore constructed. In fact, the external diameter of the Cyclone is no greater than that of the Wright Whirlwind 200 hp. radial engine, which, it will be seen, is a remarkable tribute to the engineering ability of the Wright Company.

A 50-hp. test has recently been completed, which was run in accordance with the Navy's Specification E-4-12. As this engine has been developed for the Bureau of Aeronautics, Navy Department, we are not at liberty at this time to give further information as to the test, except to add that the Cyclone develops well over 500 hp. at a specific weight of less than 2 lb./hp.

This Year's Air Races

The Contest Committee of the N.A.A. met with representatives of the various governmental departments, aircraft manufacturers, and other interested parties at N.A.A. headquarters on Jan. 20. The purpose of this meeting was to discuss the general program of aircraft sports to be pursued in the United States during 1925. Outside representatives in attendance were called in to give the Contest Committee the benefit of their suggestions.

On Jan. 30 the Contest Committee met to review various race suggestions and to arrive at the location of various contests where possible. The following decisions were made. The Jacques Schneider Race will be held at Baltimore, Md., as a date yet to be determined. The date however will fall between Oct. 24 and 31. Registration for the Contest returns the same as for 1924. Entries close April 1, 1925. The Pulitzer Trophy Race will be held between Sept. 17 and Oct. 3. The location has not yet been decided.

The regulations for the Pulitzer Trophy Race will remain the same as for 1924 with the exception, however, that foreign entries may not send their wing models to an approved board in their respective countries, instead of sending their wing models to N.E.Y. (or test, as was required last year).

In view of the fact that France has taken the world's high speed record from this country, it appears very possible that they will send their fast planes to America in complete for the Pulitzer Trophy this year. It is doubtful if their present planes can meet the landing speed requirements, but information obtained from authoritative sources indicates that requirement can be made to install larger wings and otherwise reduce their speed planes.

In view of the fact that this is an international competition, it has been necessary to close entries for the Pulitzer as April 1, and

The Committee after very careful deliberation decided to place the engine limits for single motor light planes at 50 hp. In view of the fact that development has not progressed far enough, the Committee did not feel it desirable to place limits for two-place light planes, although they passed a recommendation that 100 hp. in single motors favor the consideration of the contest. It was

Other reports were reviewed and some modifications made in the 1924 regulations.

National Balloon Race

Preliminary arrangements have been made by the National Aeronautic Association with St. Joseph, Mo., to conduct the National Klemmerton Balloon Race from that city on May 3, next. This race will be for 150 balloons at the land, fourth, and fifth categories, inclusive, (for \$1,000 or 2, to \$1,500 or 3, to \$2,000 or 4, to \$2,500 or 5, to \$3,000 or 6, to \$3,500 or 7, to \$4,000 or 8, to \$4,500 or 9, to \$5,000 or 10, to \$5,500 or 11, to \$6,000 or 12, to \$6,500 or 13, to \$7,000 or 14, to \$7,500 or 15, to \$8,000 or 16, to \$8,500 or 17, to \$9,000 or 18, to \$9,500 or 19, to \$10,000 or 20, to \$10,500 or 21, to \$11,000 or 22, to \$11,500 or 23, to \$12,000 or 24, to \$12,500 or 25, to \$13,000 or 26, to \$13,500 or 27, to \$14,000 or 28, to \$14,500 or 29, to \$15,000 or 30, to \$15,500 or 31, to \$16,000 or 32, to \$16,500 or 33, to \$17,000 or 34, to \$17,500 or 35, to \$18,000 or 36, to \$18,500 or 37, to \$19,000 or 38, to \$19,500 or 39, to \$20,000 or 40, to \$20,500 or 41, to \$21,000 or 42, to \$21,500 or 43, to \$22,000 or 44, to \$22,500 or 45, to \$23,000 or 46, to \$23,500 or 47, to \$24,000 or 48, to \$24,500 or 49, to \$25,000 or 50, to 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WRITE FOR INFORMATION

Publisher's News Letter

So much is happening in aviation circles which ought to be fully covered in a weekly publication aiming to give to readers a record of the times that this letter to our readers is in the nature of an explanation. As we have said before, one of the difficulties which confronts a weekly publication is the selection of news that will probably not be seen when the paper reaches its subscribers. An excellent illustration of this editorial predicament is before us now. Aeronautical news appears on the first page of almost every newspaper every day. And each day brings a deluge of the same. To look ahead a week or two and guess what will be news then is frankly hopeless. If you will put yourself back a week or two days before you read this, you will help in understanding our problems.

General Mitchell is again the storm center of a controversy the intensity of which can only be compared to the Sargasso-Sealey dispute or the Johnson-Rosenberg dispute. The difference is that in this case, as with Admiral Sargasso, the controversy is that General Mitchell has once again caught the public ear with his claims for aircraft and, as a most sensational moment, shaken the whole organization of our national defense. His possible decision for outpouring criticism, his challenge to the Navy, his attack on War Department conservatism, as well as his championship of a United Air Force, have all stirred Congress and the country to demand some authoritative answer to these questions. To predict or even guess what will happen a week hence would require the quality of prophecy that only you possess.

A particularly keen observer has outlined the situation in a way which will enable you to follow the warlike controversy in Washington to a carefully planned attack by an army and navy on an independent air force under a reasonable and during leader. The General and Admiral had released their campaign along approved and tested lines. The Secretary of the Navy assumed the battle with his big gun supported by the President. Thus the larger scale forces came into action, with the military ready to occupy the captured positions. It was all according to precedent and tradition. But suddenly the air force went into action with gas bombs against the capital and the civil population, as well as laying a smoke screen before the enemy. As quickly as would happen in war, the civilians were involved and the well planned attack of the surface forces was hopelessly groping in a fog of enveloping publicity. The naval and military forces pressed against the flank movement to bring control to all precedent and regulations, but the more they pressed, the faster the gas bombs of publicity were let loose from the aircraft. The outcome of the struggle cannot be forecast. America claims that a sudden aerial attack may settle the issue of the future.

It is just possible that even though there may be casualties in both planes, the surprise attack may have so startled the civilian population that the military and naval forces will be called back.

To translate this warlike portrayal into plain language, the situation at this time seems to be developing into a no-man's-land demand for a national air policy. General Mitchell has, completely upending the consequences to himself and his career, stated the nation's point of view so forcefully that the country is awaiting the reply of the bureaucracy. The situation is an attitude of mind that is demanding convincing information of the statements made regarding the power of aircraft. Coming, as it will have to, after the revolution that \$455,000,000 has been spent on an agreement between the navy and the army, it is not to be spent on aircraft current, there will have to be a radical change in the way that will be revolutionary. The naval forces, though the military leadership, have been placed in a very difficult position. With Secretary Wilson's positive assertion that a 2,000 pound bomb would not even "jar the turret" of a battleship and his further claim "We know it will not," that it has been tried and that such a statement is "absolutely unscientific and ridiculous," the Navy is placed in an untenable position. Photographs in numerous papers show the aircraft in the air. The public has seen them and will place the burden of aerial demonstration on the Navy.

Then the conflicting statements given out by the Navy concerning the bombing of the battleship Washington will require more explaining. The Navy says to the press the impression that "no less than 30,000 pounds of explosives were dropped on her" and that "no naval plane is carrying a bomb 1,600 feet" would attack the battleship. Now it appears that "no bomb of any description has been released from airplanes in flight on the completed aircraft-carrier." Such conflicting assertions will have to be unambiguously explained. The report of the General Board of the Navy will undoubtedly state the naval utility of aircraft but with the above two statements in mind the public will be bewildered.

Whether General Mitchell is devoted, count me in, or uncommitted, is beside the point of the problem. The public will not be satisfied with any course except that aircraft proponents either be given a fair chance to prove their claims or that the older services permit aviation to play a more important role in our national defense. President Coolidge can be asked to take a broad view of this controversy and may settle it in his own way, possibly as President Roosevelt settled the Admiral Sargasso—L. D. G.

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